

Gravitational force and acceleration

Force of gravity between two masses

$$F = G \frac{m_1 m_2}{r^2}$$

F - Force, N; m_1, m_2 - mass, kg; M - mass of primary, kg; r - distance between mass' centers, m

Gravitational Field Strength (Acceleration)

$$g = G \frac{M}{r^2}$$

Gravitational Constant

$$G = 6.67 \times 10^{-11}$$

Gravitational Potential Energy and Potential

Gravitational Potential Energy

$$U = -G \frac{m_1 m_2}{r}$$

U - Gravitational PE, J; V - Gravitational Potential, J/kg; m_1, m_2 - mass, kg; M - mass of primary, kg; r - distance between mass' centers, m

Gravitational Potential

$$V = -G \frac{M}{r}$$

Orbits

Orbital Velocity

$$V_{orb} = \sqrt{G \frac{M}{r}}$$

Orbital Period

$$T = \frac{2\pi r}{V_{orb}} \quad T^2 = \frac{4\pi^2 a^3}{GM}$$

Total Orbital Energy (PE + KE)

$$E = -G \frac{Mm}{2a}$$

V_{orb} - Orbital velocity, m/s; T - orbital period, sec; M, m - mass of primary, secondary, kg; r - distance to primary center, m; a - semi-major axis, m

Miscellaneous

Escape Velocity

$$V_{esc} = \sqrt{2G \frac{M}{r}}$$

$$V_{esc} = \sqrt{2} V_{orb}$$

Kepler's Third Law: Orbital Period vs Radius

$$T^2 = \frac{4\pi^2}{GM} a^3$$

For solar orbits: $\frac{T^2}{R^3} = 1$

V_{esc} - Escape velocity, m/s; M - mass of primary, kg; r - distance to center, m; T - period; R - orbital radius; a - semi-major axis

Note that this works only if T is in earth years and R is in a.u.